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## FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL HARDWARE

NUMBER: 05-6J-2033 -X

SUBSYSTEM NAME: EPD&C - MAIN PROPULSION SYSTEM

**REVISION:** 0 07/27/00

## **PART DATA**

PART NAME PART NUMBER
VENDOR NAME VENDOR NUMBER

LRU : AFT PCA 4, 5, 6 V070-765280

SRU: RELAY, GENERAL PURPOSE MC455-0129-0001

#### **EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:**

RELAY, GENERAL PURPOSE, GH2/GO2 FLOW CONTROL VALVE (LV53/54/55/56/57/58) CLOSE SOLENOID.

**REFERENCE DESIGNATORS**: 54V76A134K1

55V76A134K2 56V76A135K1 54V76A135K2 54V76A136K1 54V76A136K2

**QUANTITY OF LIKE ITEMS:** 6

#### **FUNCTION:**

PROVIDES OUPUT TO SIGNAL CONDITIONER FROM EITHER PRIMARY OR STANDBY ULLAGE PRESSURE TRANSDUCER.

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## FAILURE MODES EFFECTS ANALYSIS FMEA -- CIL FAILURE MODE

NUMBER: 05-6J-2033-03

**REVISION#:** 0 01/17/01

SUBSYSTEM NAME: EPD&C - MAIN PROPULSION SYSTEM

LRU: AFT PCA 4, 5, 6 CRITICALITY OF THIS ITEM NAME: GH2/GO2 ULLAGE PRESSURE CONTROL RELAY FAILURE MODE: 1R2

**FAILURE MODE:** 

LOSS OF OUTPUT, SHORTS TO GROUND

MISSION PHASE: LO LIFT-OFF

**VEHICLE/PAYLOAD/KIT EFFECTIVITY:** 102 COLUMBIA

103 DISCOVERY104 ATLANTIS105 ENDEAVOUR

CAUSE:

PIECE PART FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK, PROCESSING ANOMALY.

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

**REDUNDANCY SCREEN** A) PASS

B) FAIL

C) PASS

PASS/FAIL RATIONALE:

A)

B)
FAILS B SCREEN BECAUSE NO INSTRUMENTATION IS AVAILABLE TO DETECT FAILURE.

C)

## - FAILURE EFFECTS -

### (A) SUBSYSTEM:

INADVERTENT SWITCHING OF NO. 4 TRANSDUCER.

## (B) INTERFACING SUBSYSTEM(S):

NO EFFECT - FIRST FAILURE.

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## FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL FAILURE MODE NUMBER: 05-6J-2033-03

### (C) MISSION:

NO EFFECT - FIRST FAILURE.

#### (D) CREW, VEHICLE, AND ELEMENT(S):

NO EFFECT - FIRST FAILURE.

## (E) FUNCTIONAL CRITICALITY EFFECTS:

CASE 1: LH2 FAILURES.

1R/2 2 SUCCESS PATHS. TIME FRAME - ASCENT.

- 1) RELAY FAILS CAUSING LOSS OF OUTPUT TO SIGNAL CONDITIONER, LOSS OF POWER TO FCV AND FCV CYCLING TO HIGH POSITION.
- 2) INADVERTENT DEACTUATION OF SECOND FLOW CONTROL VALVE.

RESULTS IN EXCESSIVE ULLAGE PRESSURE CAUSING ET VENT VALVE TO RELIEVE EXCESS PRESSURE. POTENTIAL FIRE/EXPLOSION HAZARD EXTERIOR TO THE VEHICLE. POSIBLE VIOLATION OF THE ET MAXIMUM STRUCTURAL CAPABILITY REQUIREMENTS. POSSIBLE LOSS OF CREW/VEHICLE.

CASE 2: LO2 FAILURES. (ACTIVE GO2 VALVES ONLY) 1R/3 3 SUCCESS PATHS. TIME FRAME - ASCENT.

- 1) RELAY FAILS CAUSING LOSS OF OUTPUT TO SIGNAL CONDITIONER, LOSS OF POWER TO FCV AND FCV CYCLING TO HIGH POSITION.
- INADVERTENT DEACTUATION OF SECOND FLOW CONTROL VALVE.
- 3) FAILURE OF ET LO2 VENT VALVE TO RELIEVE.

EXCESSIVE LO2 ULLAGE PRESSURE MAY EXCEED LO2 TANK STRUCTURAL LIMITS. FIRE/EXPLOSION HAZARD. POSSIBLE LOSS OF CREW/VEHICLE.

#### -DISPOSITION RATIONALE-

# (A) DESIGN:

REFER TO APPENDIX C, ITEM NO. 2 - GENERAL PURPOSE RELAY.

#### (B) TEST:

REFER TO APPENDIX C, ITEM NO. 2 - GENERAL PURPOSE RELAY.

**GROUND TURNAROUND TEST** 

ANY TURNAROUND CHECKOUT IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD.

# (C) INSPECTION:

REFER TO APPENDIX C, ITEM NO. 2 - GENERAL PURPOSE RELAY.

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## FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL FAILURE MODE NUMBER: 05-6J-2033-03

#### (D) FAILURE HISTORY:

REFER TO APPENDIX C, ITEM NO. 2 - GENERAL PURPOSE RELAY.

CURRENT DATA ON TEST FAILURE, FLIGHT FAILURE, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATABASE.

## (E) OPERATIONAL USE:

NO CREW ACTION CAN BE TAKEN FOR FAILURES WHICH RESULT IN EXCESSIVE GH2/GO2 ULLAGE PRESSURE.

#### - APPROVALS -

S&R ENGINEERING : W.P. MUSTY :/S/ W.P. MUSTY

S&R ENGINEERING ITM : P. A. STENGER-NGUYEN :/S/ P.A. STENGER-NGUYEN

DESIGN ENGINEERING : ANDY RIZVI :/S/ ANDY RIZVI
MPS SUBSYSTEM MGR. : TIM REITH :/S/ TIM REITH
EPD&C SUBSYSTEM MGR. : RICHARD PHAN :/S/ RICHARD PHAN
MOD : JEFF MUSLER :/S/ JEFF MUSLER
USA SAM : MICHAEL SNYDER :/S/ MICHAEL SNYDER
USA ORBITER ELEMENT : SUZANNE LITTLE :/S/ SUZANNE LITTLE
NASA SR&QA : BILL PRINCE :/S/ BILL PRICE